







## Recap: Non-plane case, solution

$$V_\nu(u, v, w) = \iint \frac{I_\nu(l, m)}{\sqrt{1 - l^2 - m^2}} e^{-i2\pi(ul+vm+wn)} dl dm \quad (1)$$

General form is not 3D FT, using small angle approximation we get:

$$V_\nu(u, v, w) = \iint \frac{I_\nu(l, m)}{\sqrt{1 - l^2 - m^2}} e^{-i2\pi(ul+vm+w\theta^2/2)} dl dm \quad (2)$$

If  $w\theta^2 \ll 1$  or  $\theta_{max} \lesssim \sqrt{\theta_{synth}}$ , it can be neglected. E.g. for a 0.1 milliarcsecond synthesized beam the limit would be about 144 milliarcseconds. Then further

$$V'_\nu(u, v) = \iint \frac{I_\nu(l, m)}{\sqrt{1 - l^2 - m^2}} e^{-i2\pi(ul+vm)} dl dm, \quad (3)$$

which is again a 2D FT, yippee!



**Baseline** sensitivity for antennas  $i$  and  $j$  ( $\eta_s =$  system efficiency):

$$\Delta S_{ij} = \frac{1}{\eta_s} \sqrt{\frac{\text{SEFD}_i \cdot \text{SEFD}_j}{2\Delta\nu\tau_{int}}} \quad [\text{Jy}] \quad (6)$$

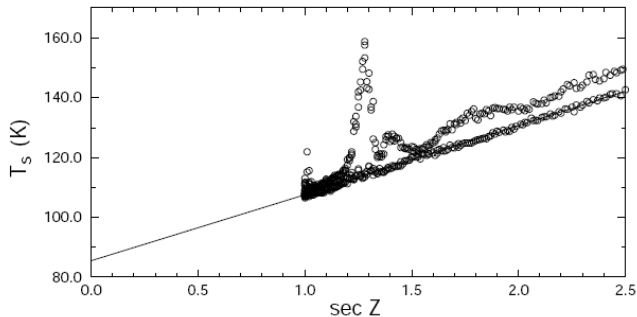
**Image** sensitivity  $I_m$  is standard deviation of mean of  $L$  samples (baselines),

$$\Delta I_m = \frac{1}{\eta_s} \sqrt{\frac{\text{SEFD}_i \cdot \text{SEFD}_j}{N(N-1)\Delta\nu\tau_{int}}} \quad [\text{Jy/beam}] \quad (7)$$



# Recap: Atmospheric opacity

$T_{\text{sys}}$  plotted as a function of airmass (  $T_{\text{sys}} \simeq T_R + T_0 \tau_0 \sec z$  ):



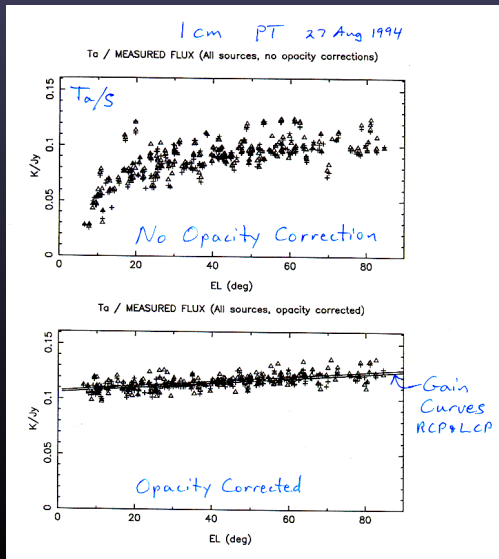
Zenith opacity  $\tau_0$  is the slope of the linear fit (times  $T_0$ ) and  
 $T_R = T_{\text{sys}}(\text{airmass} = 0)$

Figure from VLBA-book

# Atmospheric opacity correction

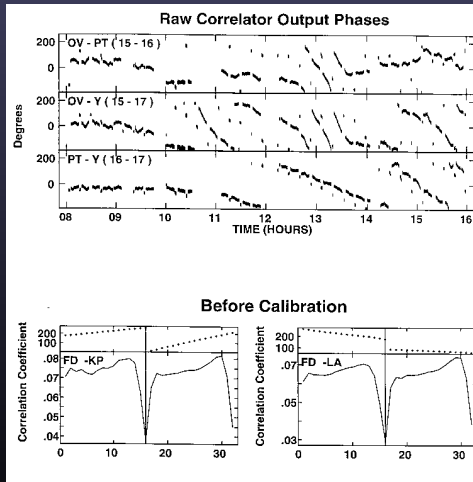
- Corrections for absorption by the atmosphere
- Can estimate using  $T_s - T_r - T_{\text{spill}}$

Example from VLBA single dish pointing data



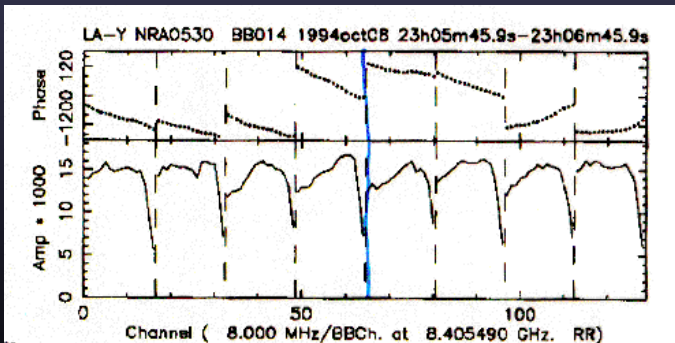
## Residual rate and delay

- Interferometer phase  $\phi_{l,v} = 2\pi\nu\tau_l$
- Slope in frequency is “delay”
  - Fluctuations worse at low frequency because of ionosphere
  - Troposphere affects all frequencies equally (“nondispersive”)
- Slope in time is “fringe rate”
  - Usually from imperfect troposphere or ionosphere model



## Instrumental delays

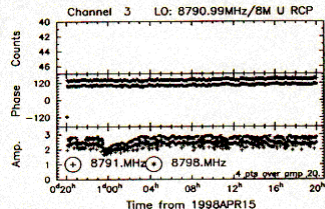
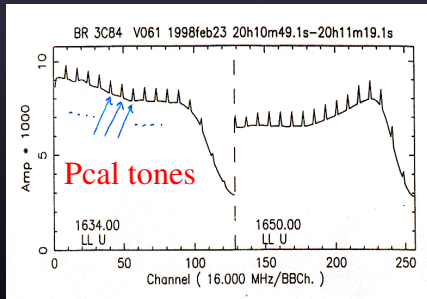
- Caused by different signals paths through the electronics in the separate bands



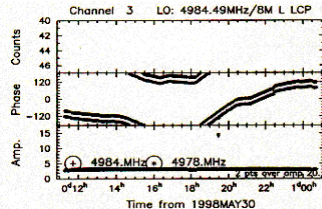


# The pulse cal

- Corrected for using the pulse cal system (continuum only)
- Tones generated by injecting a pulse every microsecond

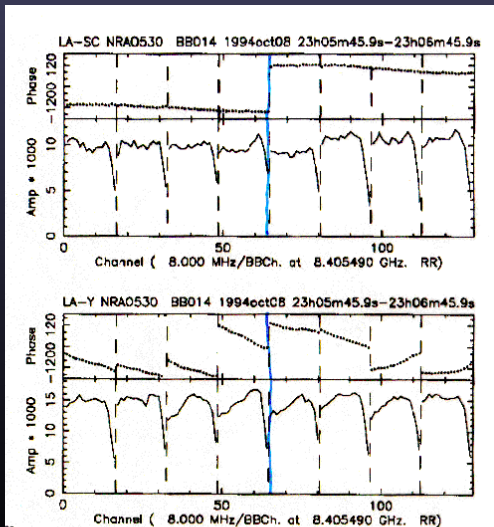


## Pulse cal monitoring data



## Corrections using Pcal

- Data aligned using Pcal
- No Pcal at VLA, shows unaligned phases



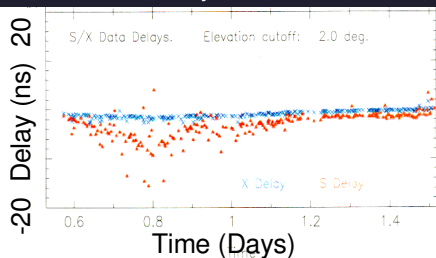
## Ionospheric delay

- Delay scales with  $1/\nu^2$
- Ionosphere dominates errors at low frequencies
- Can correct with dual band observations (S/X) or GPS based models

### Maximum Likely Ionospheric Contributions

Freq GHz	Day Delay ns	Night Delay ns	Day Rate mHz	Night Rate mHz
0.327	1100	110	12	1.2
0.610	320	32	6.5	0.6
1.4	60	6.0	2.8	0.3
2.3	23	2.3	1.7	0.2
5.0	5.0	0.5	0.8	0.1
8.4	1.7	0.2	0.5	0.05
15	0.5	0.05	0.3	0.03
22	0.2	0.02	0.2	0.02
43	0.1	0.01	0.1	0.01

### Delays from an S/X Geodesy Observation

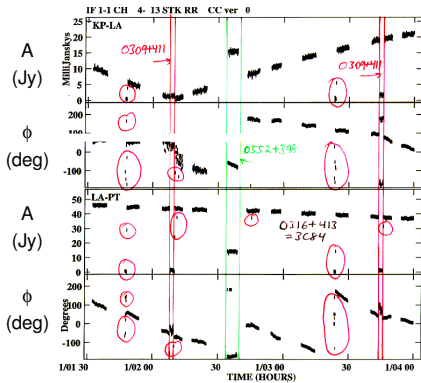


## Editing

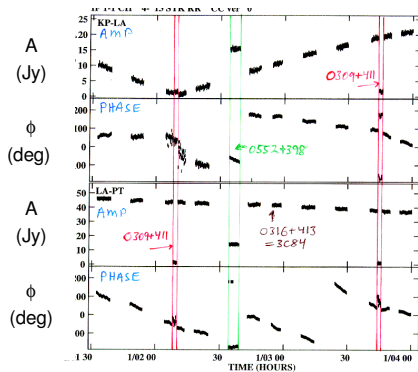
- Flags from on-line system will remove most bad data
  - Antenna off source
  - Subreflector out of position
  - Synthesizers not locked
- Final flagging done by examining data
  - Flag by antenna (most problems are antenna based)
  - Poor weather
  - Bad playback
  - RFI (may need to flag by channel)
  - First point in scan sometimes bad

# Editing example

## Raw Data - No Edits

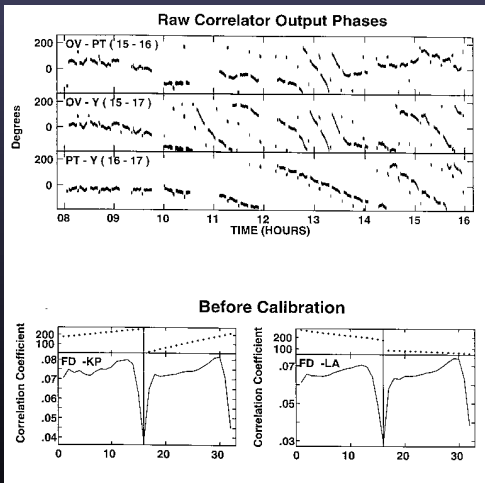


## Raw Data - Edited



## Phase errors

- Raw correlator output has phase slopes in time and frequency
- Caused by imperfect delay model
- Need to find delay and delay-rate errors



## Fringe fitting

- For astronomy:
  - Remove clock offsets and align baseband channels (“manual pcal”)
  - Fit calibrator to track most variations
  - Fit target source if strong
  - Used to allow averaging in frequency and time
    - Allows higher SNR self calibration (longer solution, more bandwidth)
- For geodesy:
  - Fitted delays are the primary “observable”
  - Correlator model is added to get “total delay”, independent of models





Phase due delay and rate:

$$\Delta\phi_{t,\nu} = \phi_0 + \left( \frac{\partial\phi}{\partial\nu}\Delta\nu + \frac{\partial\phi}{\partial t}\Delta t \right) \quad (9)$$

Baseline phase errors due to delay and rate:

$$\Delta\phi_{ij} = \phi_{i0} - \phi_{j0} + \left( \left[ \frac{\partial\phi_i}{\partial\nu} - \frac{\partial\phi_j}{\partial\nu} \right] \Delta\nu + \left[ \frac{\partial\phi_i}{\partial t} - \frac{\partial\phi_j}{\partial t} \right] \Delta t \right) \quad (10)$$

# Fringe fitting and self calibration

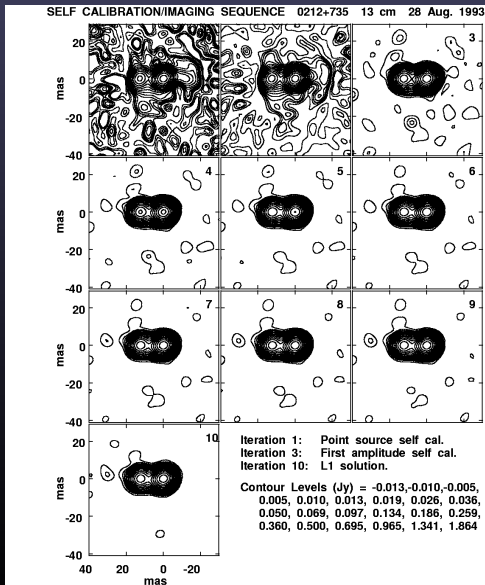
Delays, rates, and phases are usually solved for **antennas** not for baselines, this is called *global fringe fitting* and gives better sensitivity (all data for a given antenna is used). In some special cases *baseline based* fringe fitting is used.

In both methods, the source is assumed to be point-like (constant phases and amplitudes), if not, a model for the source can be used.

**Self calibration** is solving the *antenna phases* and sometimes amplitudes (not visibility phases!!) based on a source model.

# Self calibration imaging sequence

- Iterative procedure to solve for both image and gains:
  - Use best available image to solve for gains (start with point)
  - Use gains to derive improved image
  - Should converge quickly for simple sources
- Does not preserve absolute position or flux density scale







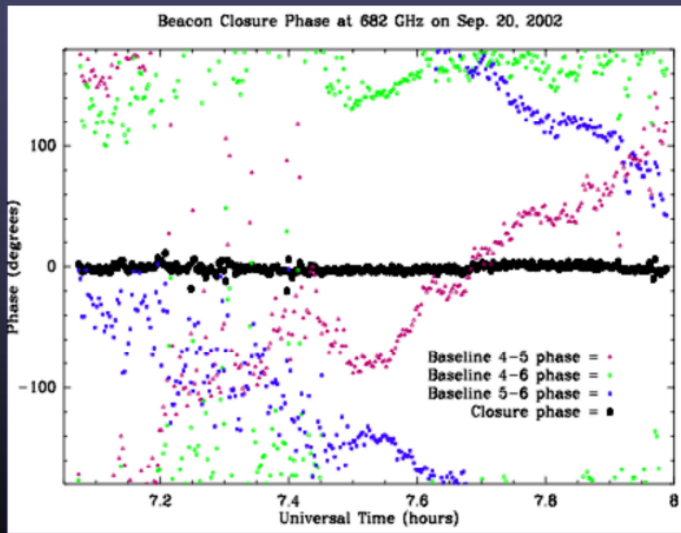








# SMA closure phase measurements at 682GHz





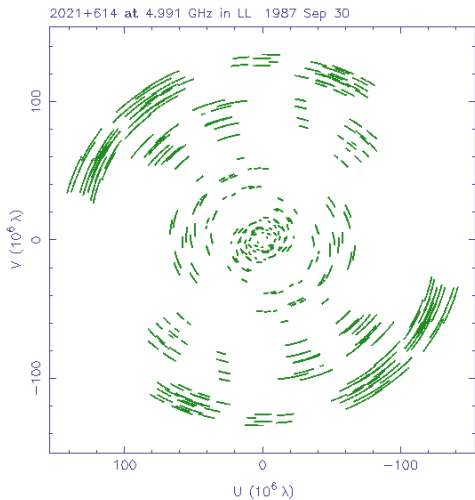
Before imaging it is very useful to make plots of visibility phase and amplitude:

- ▶ vs. uv-radius
- ▶ vs. time
- ▶ vs. uv-projection (slice across the uv-plane)

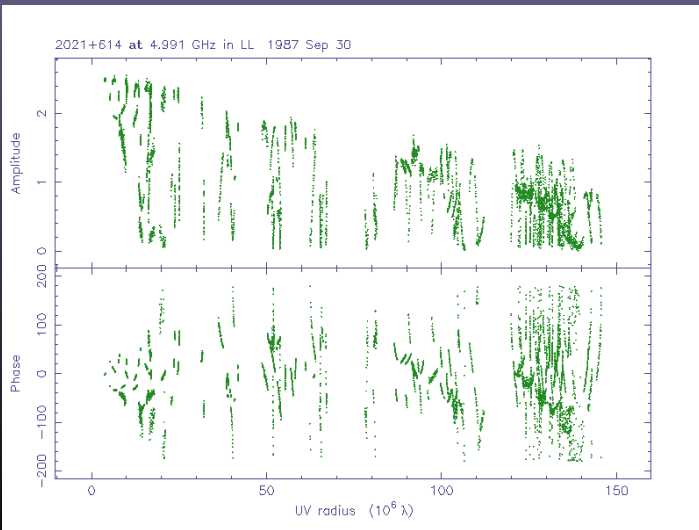
UV-coverage (sampling) map tells the general quality that is to be expected (sidelobe/artifact level).

These plots give first ideas what to expect from the source structure.

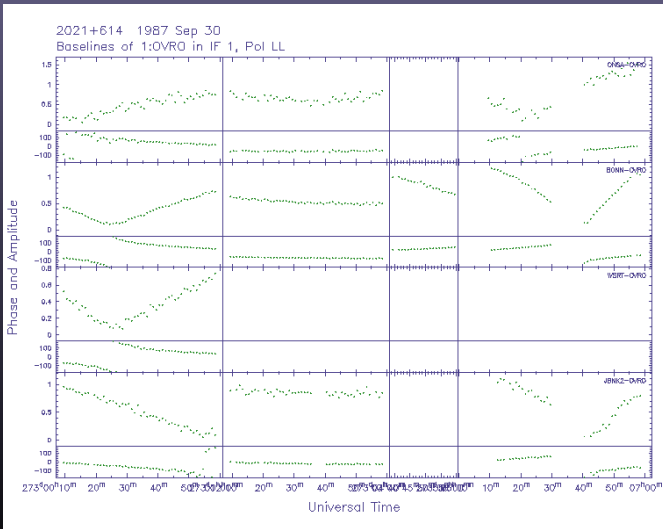
## Sampling of the (u,v) plane



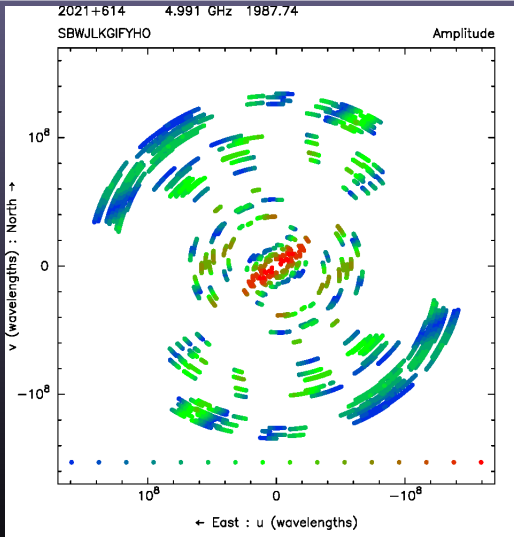
## Visibility versus (u,v) radius



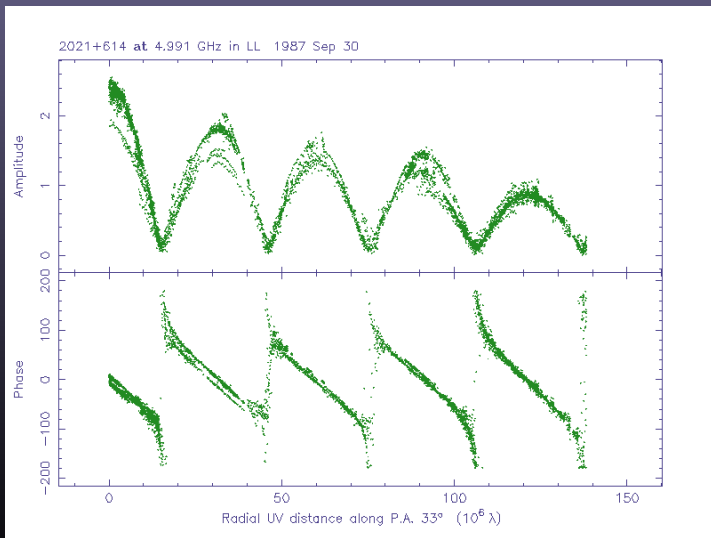
# Visibility versus time



# Amplitude across the (u,v) plane



## Projection in the (u,v) plane





# Fourier transform properties

$$F(u, v) = \text{FT}\{f(x, y)\}$$

i.e.,

$$F(u, v) = \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} f(x, y) \exp[2\pi i(ux + vy)] dx dy$$

## Linearity

$$\text{FT}\{f(x, y) + g(x, y)\} = F(u, v) + G(u, v)$$

## Convolution

$$\text{FT}\{f(x, y) \star g(x, y)\} = F(u, v) \cdot G(u, v)$$

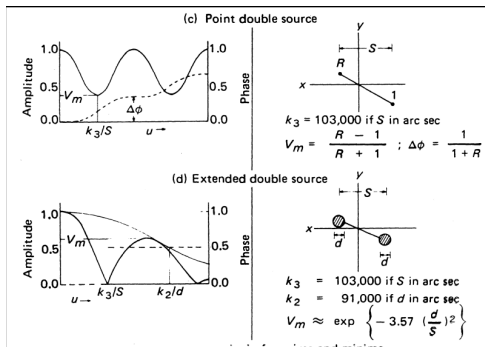
## Shift

$$\text{FT}\{f(x - x_i, y - y_i)\} = F(u, v) \exp[2\pi i(ux_i + vy_i)]$$

## Similarity

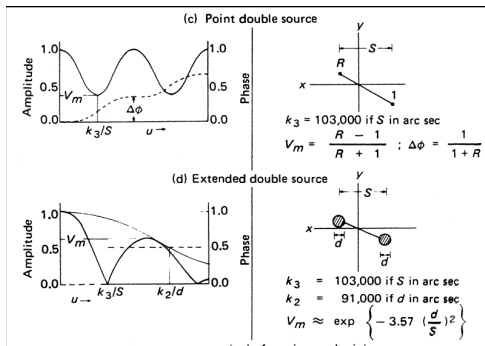
$$\text{FT}\{f(ax, by)\} = \frac{1}{|ab|} F\left(\frac{u}{a}, \frac{v}{b}\right)$$

# Simple source structures



Component separation from the  $uv$ -radius (in wavelengths) of the first valley ( $k_3/S$ ), size of individual emission region ( $d$  [arcsec]) from the  $uv$ -radius of the half-value point of the envelope ( $k_2/d$ ). Amplitude is normalized.

# Simple source structures, example



First valley at  $100 M\lambda = k_3/S$ , envelope half-value point  $300 M\lambda = k_2/d$ .

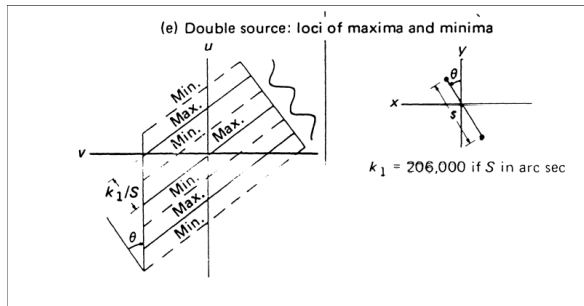
Double source, component separation

$$S = k_3/100M\lambda = 103000/100e6 = 0.001 \text{ arcsec} = 1 \text{ marcsec.}$$

Component size

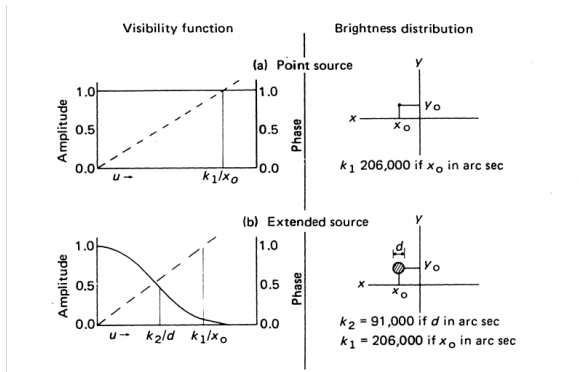
$$d = k_2/300M\lambda = 91000/300e6 = 0.0003 \text{ arcsec} = 300 \mu\text{arcsec}$$

# Simple source structures in 2D

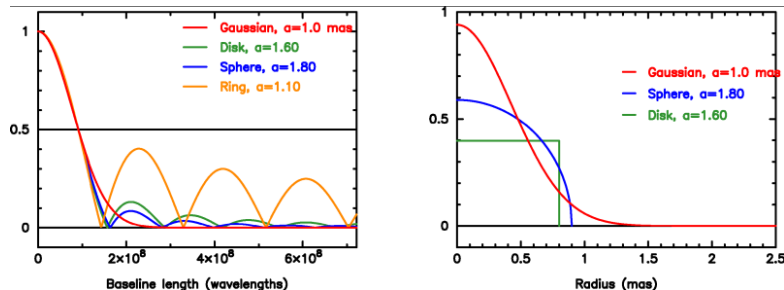


Component separation from the valley-to-valley distance ( $k_1/S$ ).

# Source shift



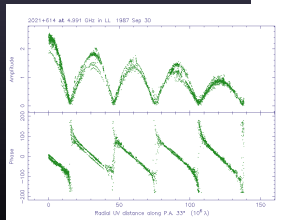
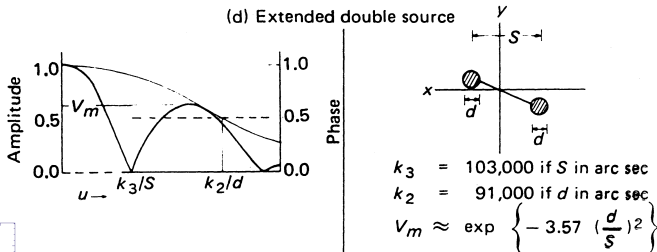
# Component profiles



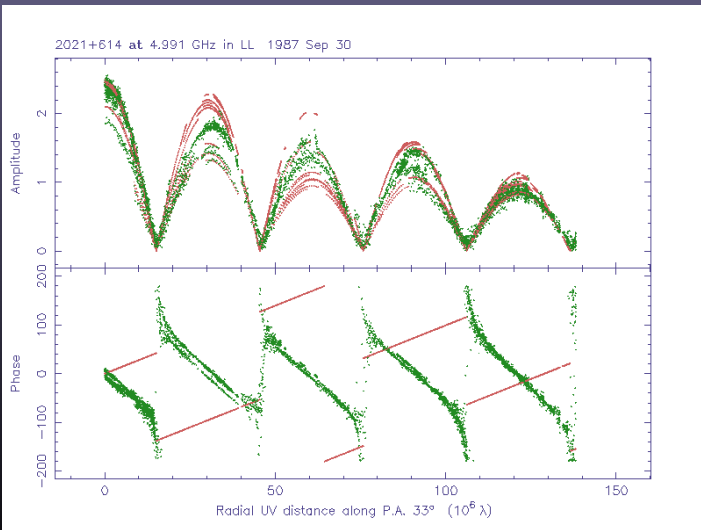
There is very little difference in the uv-plane between different source profiles down to the relative half flux level.

## Trial model

- By inspection, we can derive a simple model:
- Two equal components, each 1.25 Jy, separated by about 6.8 milliarcsec in p.a.  $33^\circ$ , each about 0.8 milliarcsec in diameter (Gaussian FWHM)
- *To be refined later.*



## Projection in the (u,v) plane



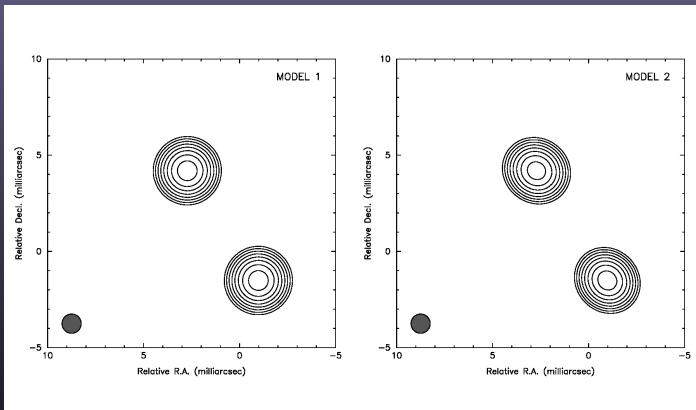


# Parameters

- Example
  - Component position:  $(x,y)$  or polar coordinates
  - Flux density
  - Angular size (e.g., FWHM)
  - Axial ratio and orientation (position angle)
    - For a non-circular component
  - 6 parameters per component, plus a “shape”
  
  - This is a conventional choice: other choices of parameters may be better!
  - (Wavelets; shapelets\* [Hermite functions])
    - \* Chang & Refregier 2002, ApJ, 570, 447

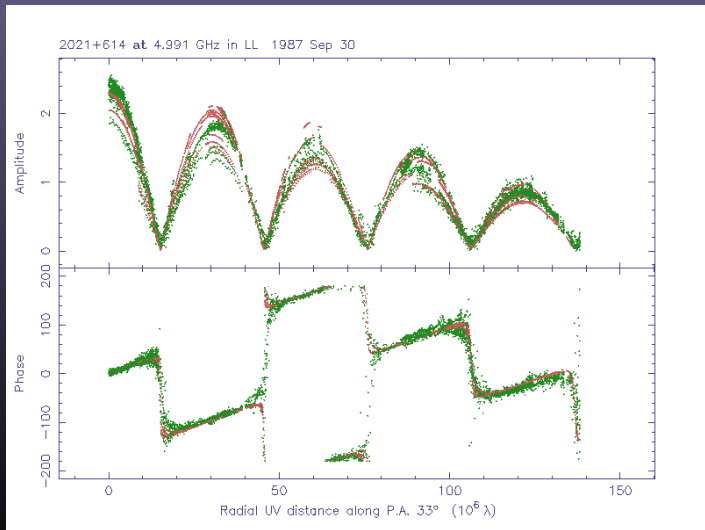


# Practical model fitting: 2021

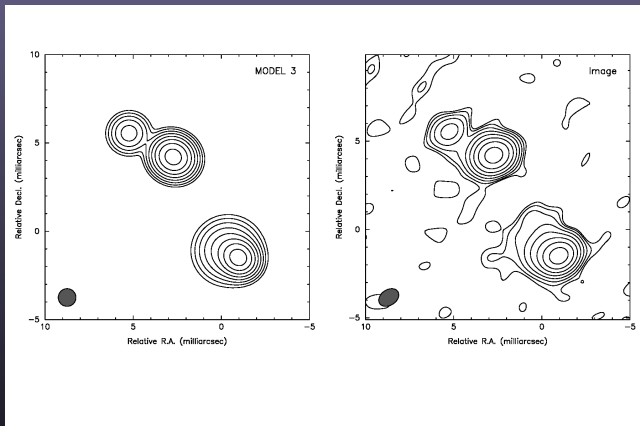


	! Flux (Jy)	Radius (mas)	Theta (deg)	Major (mas)	Axial ratio	Phi (deg)	T
•	1.15566	4.99484	32.9118	0.867594	0.803463	54.4823	1
•	1.16520	1.79539	-147.037	0.825078	0.742822	45.2283	1

## 2021: model 2

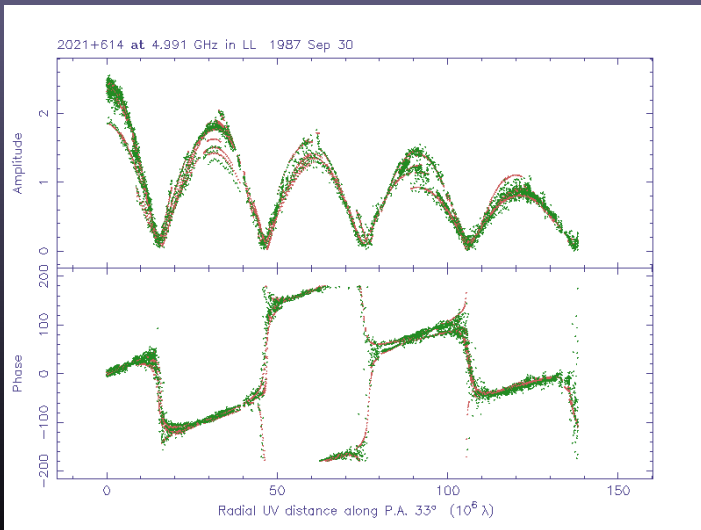


# Model fitting 2021



•	! Flux (Jy)	Radius (mas)	Theta (deg)	Major (mas)	Axial ratio	Phi (deg)	T
•	1.10808	5.01177	32.9772	0.871643	0.790796	60.4327	1
•	0.823118	1.80865	-146.615	0.589278	0.585766	53.1916	1
•	0.131209	7.62679	43.3576	0.741253	0.933106	-82.4635	1
•	0.419373	1.18399	-160.136	1.62101	0.951732	84.9951	1

## 2021: model 3



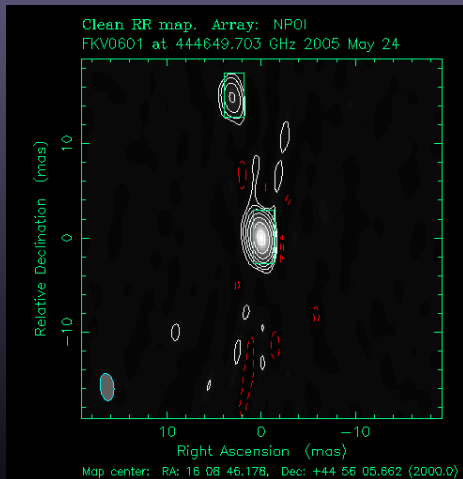
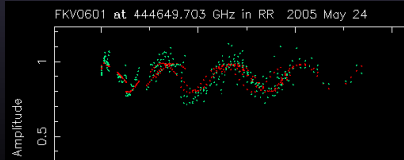
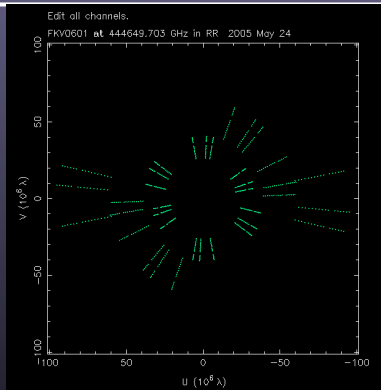
## Applications: A Binary Star

- Binary Stars
  - Many stars are in binary systems
  - Orbital parameters can be used to measure stellar masses
  - Astrometry can provide direct distances via parallax and proper motions.
- Application of model fitting
  - Optical interferometry provides sparse visibility coverage
  - Small number of components
  - Need error estimates.
- Example: NPOI observations of Phi Herculis (Zavala et al. 2006)
  - Multiple observations map out the orbit

NPOI = Navy Precision Optical Interferometer, three arm 'Y',  
250 m each.



# NPOI Observations of Phi Her









# Deconvolution/inversion methods

- ▶ Maximum Entropy Method
  - ▶ Tries to find an image that is consistent with data but has the maximum amount of entropy, i.e. smoothest image possible.
  - ▶ Does well with large scale diffuse emission, point sources are a problem.
- ▶ Variants of clean-algorithm:
  - ▶ Basic clean: scaled dirty beams are subtracted from the image plane until only residuals left. The positions and amplitudes are registered and these are replaced by a sum of Gaussians (clean-beams) with the same amplitudes.
  - ▶ Scale sensitive clean-variants: dirty beams are convolved with well-behaved functions (Gaussians, Hermitians etc.) with a number of sizes (scales) before clean-subtraction is done. The sum of the 'clean', unconvolved functions is the resulting image.
- ▶ Compressed sensing direct inversion
  - ▶ Based on a recently discovered algorithm utilizing **sparsity** of data in some domain and minimization of the L1-norm.
  - ▶ Prerequisites of convergence can be mathematically proved. ▶