

















# Antenna temperature due to a flux density

Finally we get,

$$S = \frac{2kT_a}{A_e} \left[ \frac{\text{W}}{\text{m}^2 \text{ Hz}} \right] \quad (9)$$

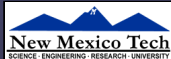
where  $T_a$  is the *antenna temperature*, i.e. the increase of the temperature of a 'resistor' due to the source flux density.

A convenient measure of antenna performance is DPFU (Degrees Per Flux Unit):

$$DPFU = \frac{A_e}{2k} \cdot 10^{-26} \left[ \frac{\text{K}}{\text{Jy}} \right] \quad (10)$$

e.g.

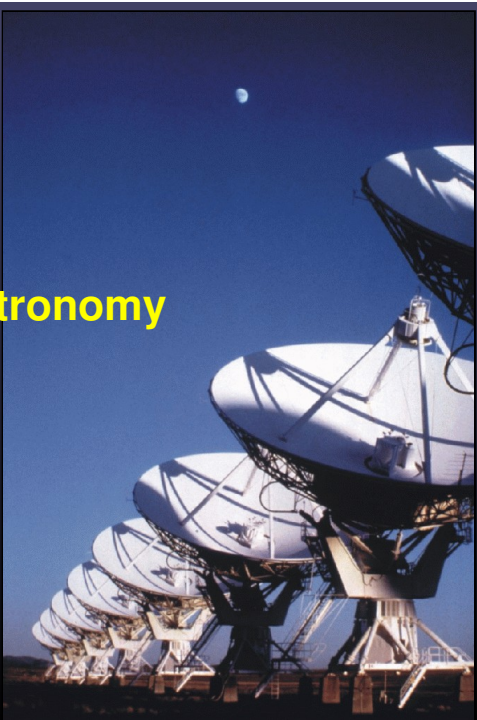
Telescope	diameter [m]	DPFU [mK/Jy]
Tuorla solar telescope	2	0.6
Metsähovi	13.7	26.7
Effelsberg	100	1580



# Antennas in Radio Astronomy

Peter Napier

*Ninth Synthesis Imaging Summer School  
Socorro, June 15-22, 2004*



# General Antenna Types

Wavelength > 1 m (approx)

Wire Antennas

 Dipole

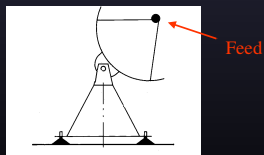
Yagi



Helix  
or arrays of these

Wavelength < 1 m (approx)

Reflector antennas

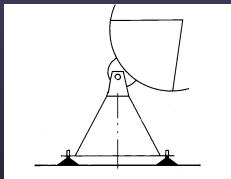


Wavelength = 1 m (approx) Hybrid antennas (wire reflectors or feeds)



## Importance of the Antenna Elements

- Antenna amplitude pattern causes amplitude to vary across the source.
- Antenna phase pattern causes phase to vary across the source.
- Polarization properties of the antenna modify the apparent polarization of the source.
- Antenna pointing errors can cause time varying amplitude and phase errors.
- Variation in noise pickup from the ground can cause time variable amplitude errors.
- Deformations of the antenna surface can cause amplitude and phase errors, especially at short wavelengths.



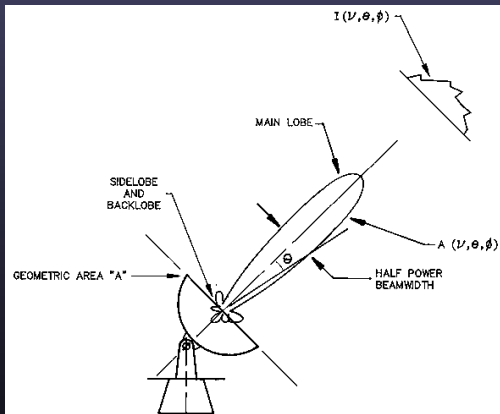
## Basic Antenna Formulas

Effective collecting  
area  $A(\nu, \theta, \phi)$   $\text{m}^2$

On-axis response  $A_0 = \eta A$   
 $\eta$  = aperture efficiency

Normalized pattern  
(primary beam)

$$A(\nu, \theta, \phi) = A(\nu, \theta, \phi) / A_0$$



Beam solid angle  $\Omega_A = \iint_{\text{all sky}} A(\nu, \theta, \phi) d\Omega$

$\nu$  = frequency

$\lambda$  = wavelength



# Aperture-Beam Fourier Transform Relationship

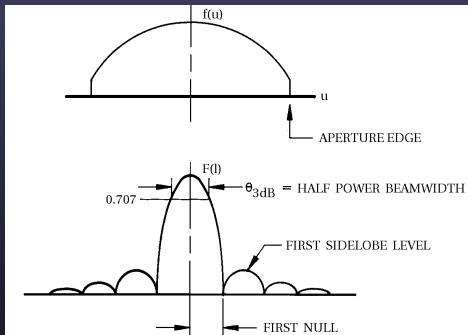
$f(u,v)$  = complex aperture field distribution  
 $u,v$  = aperture coordinates (wavelengths)

$F(l,m)$  = complex far-field voltage pattern  
 $l = \sin\theta\cos\phi$  ,  $m = \sin\theta\sin\phi$

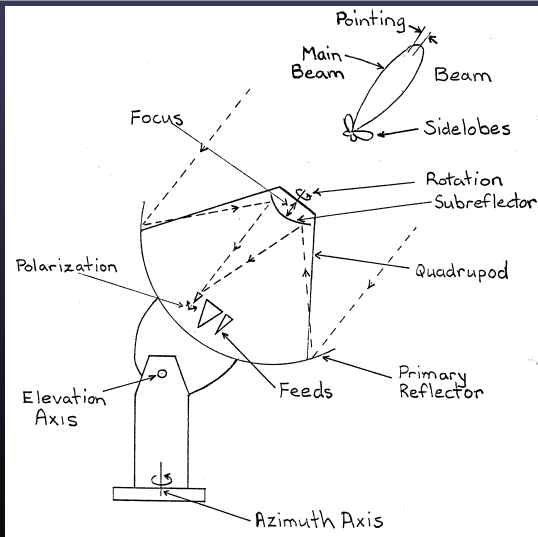
$$F(l,m) = \iint_{\text{aperture}} f(u,v) \exp(2\pi i(ul+vm)) du dv$$

$$f(u,v) = \iint_{\text{hemisphere}} F(l,m) \exp(-2\pi i(ul+vm)) dl dm$$

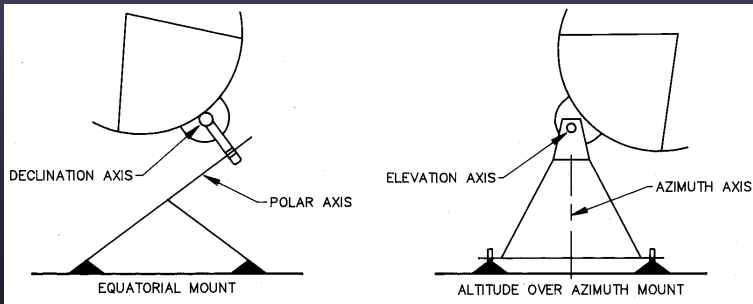
For VLA:  $\theta_{3dB} = 1.02/D$ , First null =  $1.22/D$ ,  
 $D$  = reflector diameter in wavelengths



# Primary Antenna Key Features



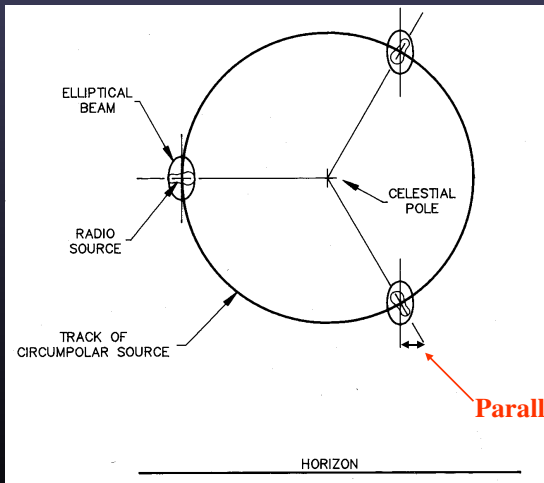
## Types of Antenna Mount



- + Beam does not rotate
- + Better tracking accuracy
- Higher cost
- Poorer gravity performance
- Non-intersecting axis

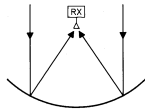
- + Lower cost
- + Better gravity performance
- Beam rotates on the sky

# Beam Rotation on the Sky

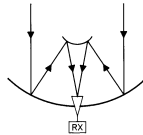


# Reflector Types

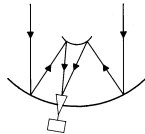
Prime focus  
(GMRT)



Cassegrain focus  
(AT)

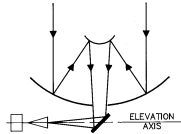


Offset Cassegrain  
(VLA)

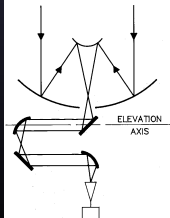


Smith

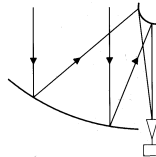
(OVRO)



Beam Waveguide  
(NRO)



Dual Offset  
(ATA)



# Reflector Types

Prime focus  
(GMRT)



Cassegrain focus  
(AT)

Offset Cassegrain  
(VLA)



Offset Cassegrain  
(OVRO)

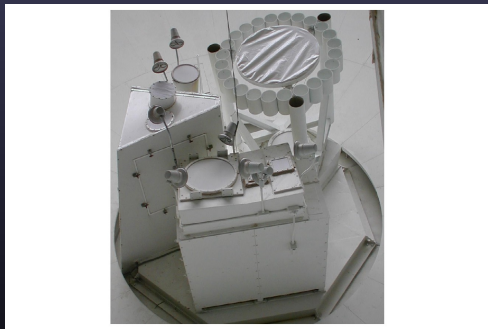
Beam Waveguide  
(NRO)



Dual Offset  
(ATA)



# VLA and EVLA Feed System Design



# Antenna Performance Parameters

## Aperture Efficiency

$$A_0 = \eta A, \eta = \eta_{sf} \times \eta_{bl} \times \eta_s \times \eta_t \times \eta_{misc}$$

$\eta_{sf}$  = reflector surface efficiency

$\eta_{bl}$  = blockage efficiency

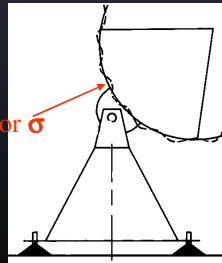
$\eta_s$  = feed spillover efficiency

$\eta_t$  = feed illumination efficiency

$\eta_{misc}$  = diffraction, phase, match, loss rms error  $\sigma$

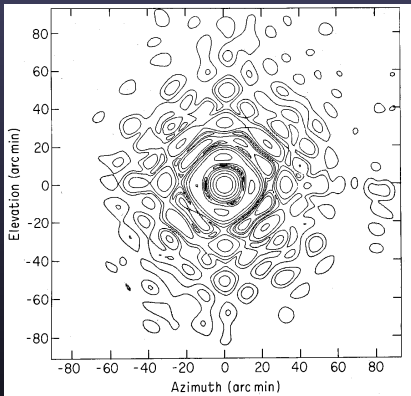
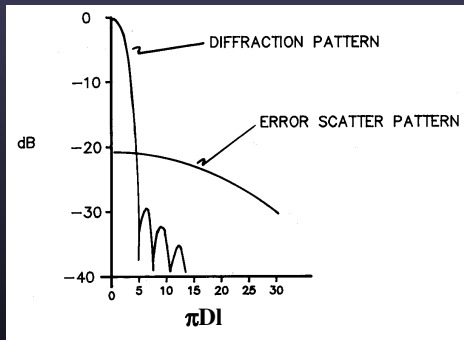
$$\eta_{sf} = \exp(-(4\pi\sigma/\lambda)^2)$$

e.g.,  $\sigma = \lambda/16$ ,  $\eta_{sf} = 0.5$



# Antenna Performance Parameters

## Primary Beam



contours: -3, -6, -10, -15, -20, -25,  
-30, -35, -40 dB

$l = \sin(\theta)$ ,  $D$  = antenna diameter in wavelengths

$\text{dB} = 10\log(\text{power ratio}) = 20\log(\text{voltage ratio})$

For VLA:  $\theta_{3\text{dB}} = 1.02/D$ , First null =  $1.22/D$

# Antenna Performance Parameters

## Pointing Accuracy

$\Delta\theta$  = rms pointing error

Often  $\Delta\theta < \theta_{3dB} / 10$  acceptable

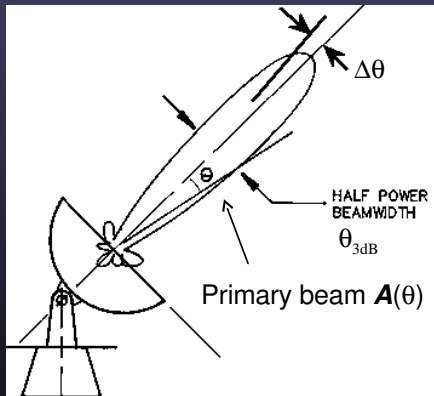
Because  $A(\theta_{3dB} / 10) \sim 0.97$

BUT, at half power point in beam

$$A(\theta_{3dB} / 2 \pm \theta_{3dB} / 10) / A(\theta_{3dB} / 2) = \pm 0.3$$

For best VLA pointing use Reference Pointing.

$$\Delta\theta = 3 \text{ arcsec} = \theta_{3dB} / 17 @ 50 \text{ GHz}$$



# Antenna Pointing Design

Reflector structure

Subreflector mount

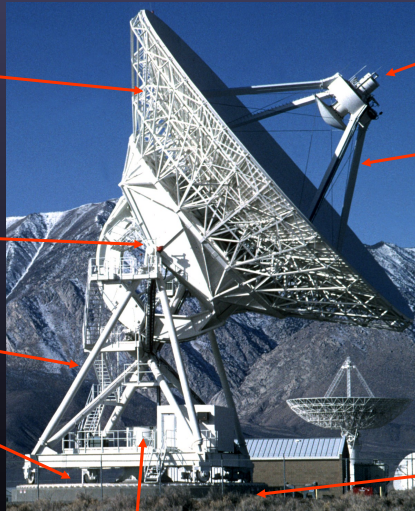
EI encoder

Quadrupod

Alidade structure

Rail flatness

Foundation



Az encoder



## ALMA 12m Antenna Design

Surface:  $\sigma = 25 \mu\text{m}$

Pointing:  $\Delta\theta = 0.6 \text{ arcsec}$

Carbon fiber and invar  
reflector structure

Pointing metrology structure  
inside alidade

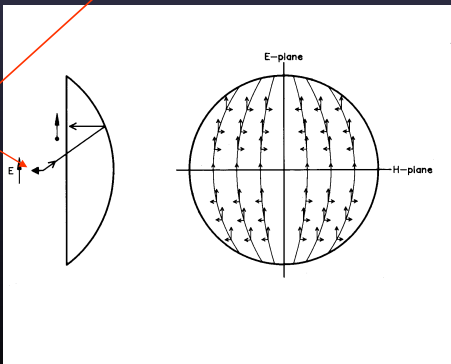
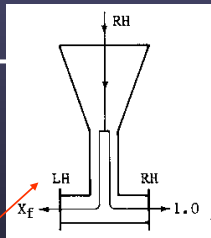


# Antenna Performance Parameters

## Polarization

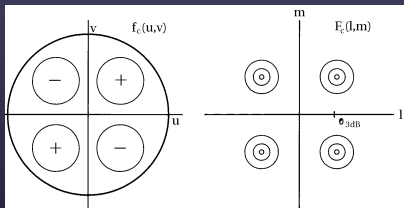
Antenna can modify the apparent polarization properties of the source:

- Symmetry of the optics
- Quality of feed polarization splitter
- Circularity of feed radiation patterns
- Reflections in the optics
- Curvature of the reflectors



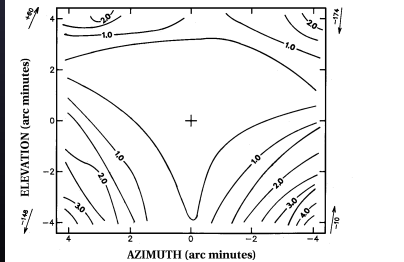
# Off-Axis Cross Polarization

Cross polarized  
aperture distribution



Cross polarized  
primary beam

VLA 4.8 GHz  
cross polarized  
primary beam



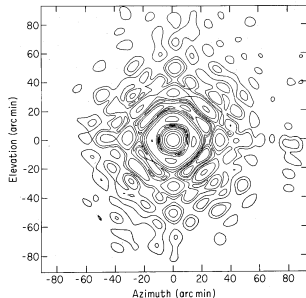


# Antenna Holography

VLA 4.8 GHz

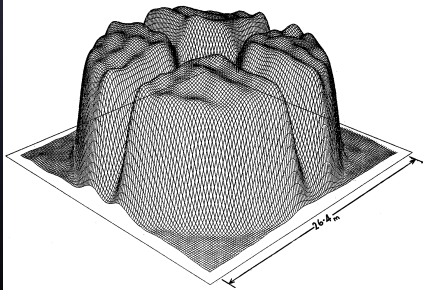
Far field pattern amplitude

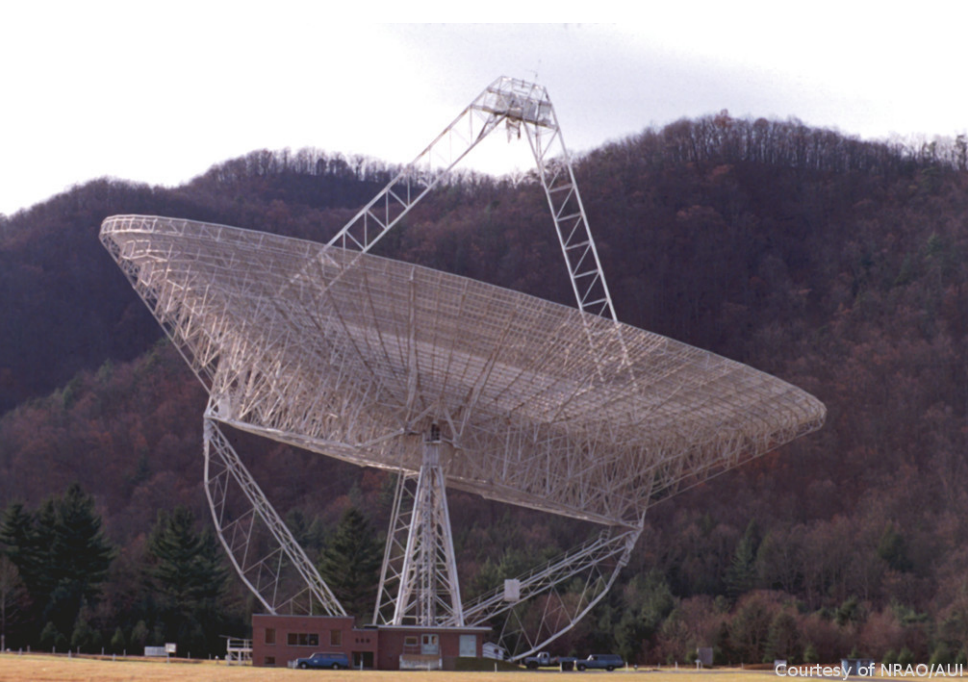
Phase not shown



Aperture field distribution  
amplitude.

Phase not shown





Courtesy of NRAO/AUI



Courtesy of NRAO/AUI

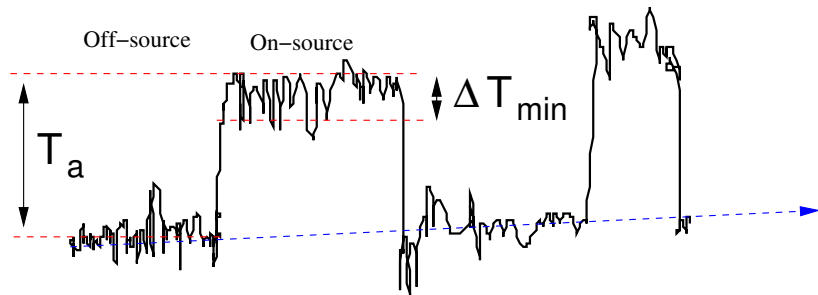


# Radiometer sensitivity

We know now that the noise temperature at our receiver input is

$$T_a = \frac{SA_e}{2k} \quad (11)$$

How much does the receiver generate noise by itself ( $T_r$ ) and what is the minimum detectable  $T_a$ ?



Here the antenna has switched between the source and a blank sky, note the increase in noise level ( $T_a$ ).

$$T_{sys} = T_a + T_r + T_{sky}$$

When determining the sensitivity of a radiometer, total noise contribution  $T_{sys} = T_a + T_r + T_{sky}$  to the radiometer must be taken into account, i.e the noise from source, receiver and atmosphere.

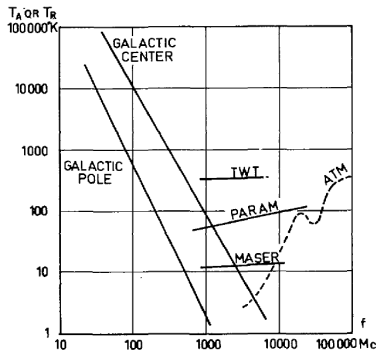


Fig. 2—Antenna sky noise temperature  $T_A$  with beam pointing at the galactic pole and galactic center and due to the atmosphere (dashed) as a function of the frequency. Noise temperatures  $T_R$  of some typical receivers are also shown.

# Total system noise temperature

It should be noted that the quantity radiometer is measuring (in radio astronomy) is noise itself and adds to the total noise temperature and to the standard deviation of the result. Because of this, *all* noise sources including the source that is been measured (antenna temperature), atmosphere, CMB and the telescope itself must be taken into account:

$$T_{\text{sys}} = T_{\text{a}} + T_{\text{atm}} + T_{\text{cmb}} + T_{\text{tel}} + \dots \quad (12)$$

The total noise of a receiver is called the *total system noise temperature*  $T_{\text{sys}}$ .

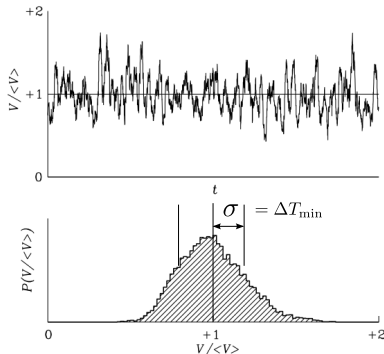
Sensitivity is also a function of *bandwidth*  $\Delta\nu$  or  $B$  from where the noise power is collected and amount of averaging, i.e. the *integration time*  $\tau$ .

# Minimum detectable noise difference $\Delta T_{min}$

Then the minimum detectable noise difference *for ideal radiometer* is

$$\Delta T_{min} = \frac{T_{sys}}{\sqrt{\Delta \nu \tau}}. \quad (13)$$

There are effectively  $\Delta \nu$  independent noise contributions per second,  $\Delta \nu \tau$  in total. Therefore RMS deviation is  $1/\sqrt{\Delta \nu \tau}$  times the average output.





## **Bolometer: polarizer, frequency band filter and thermometer**

- ▶ Power is transformed into temperature which is measured.
- ▶ Very wideband and therefore sensitive, no spectral information
- ▶ Has to be cooled to less than 4K (superconductors)

## **Direct detecting receiver: polarizer, amplifier and power detector**

- ▶ Simple and low-cost
- ▶ Microwave amplifiers can be operated (even) at room temperature
- ▶ Wideband and sensitive but no spectral information

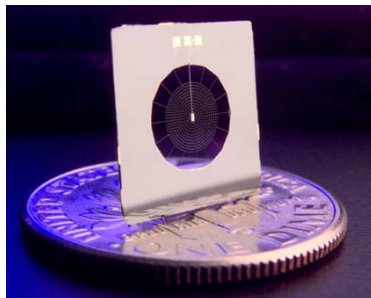
## **Heterodyne receiver**

- ▶ Signal from antenna is translated down to a more convenient frequency
- ▶ Local oscillator and mixer increase complexity
- ▶ Full spectral information available
- ▶ DSB: double (folded) sideband, SSB: single sideband

# Bolometer

A *bolometer* measures the power of incident electromagnetic radiation via the heating of a material.

To improve sensitivity, thermometers in bolometers are usually superconductors that are cooled down to below 4 K. They are very wideband instruments, that is one reason for their sensitivity.

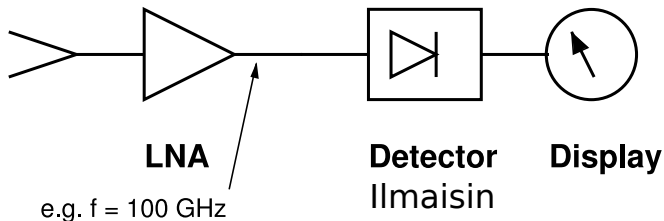


Credit: NASA/JPL-Caltech

If any frequency selectivity is needed, it must be implemented using (rather wide) filters. Because bolometers cannot measure phase of the incident radiation, they cannot be used in interferometers.

# Direct detecting receiver (or radiometer)

Direct detecting receiver consists of only amplifier and a detector. The advancement of amplifier technology in 1990's made this type of receiver more popular. Before it was necessary to downconvert the frequency before amplification.



The advantages are *simplicity* and, due to the wide bandwidth, *sensitivity*. Because phase information is lost, this receiver type cannot be used for interferometry either.

# (Super)heterodyne receiver I

In heterodyne receivers, the received frequency band is translated with a mixer to a lower, *intermediate frequency (IF)*. Especially some tens of years back, manufacturing high frequency ( $\nu > 10$  GHz) amplifiers was expensive and difficult but mixers were much easier and cheaper. Therefore downconverting in frequency and amplifying with cheaper and more manageable components (size scales with wavelength!) was considerably easier.

